



# Energy and Power Requirements for Heavy Lift Space Elevators

Up or Down We Must Get Used  
to Large Numbers

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2025 International Space Elevator Consortium

# Viable Space Settlement & Space Economy

- Initial Outbound Target Goal of 1 million KG per day
  - Support Five Basic Missions
    - Moon
    - Mars
    - Solar
    - O'Neill Colonies
    - Sunshade
  - Launch to other places in the solar system
  - Large-scale GEO space manufacturing
- Initial Earth-bound Target Goal of 1 million KG per day
  - Large-scale GEO space manufacturing
  - Bring back wealth from space (asteroid mining, etc.) to pay for settlements
  - Space Tourism

# Requirement Generation Considerations

## Goal: 1 Mkg up and down every day for Space Settlement and Economy

- Rocket-based and reentry vehicle transportation from and to Earth is inadequate for settling space and supporting a beyond LEO Space Economy
- Trillions in Rocket/Reentry vehicle Transportation Costs Require tether-based solutions with low variable cost
- Electric-based solutions appear to be the most promising for Space Elevator concepts
- How much electric power is required for single space elevator
- Does the Electric Energy and Power requirements define the solution space for number of space elevator launches and recoveries per day
- What Architecture fulfills this goal?

# Energy and Power are Required to Overcome Vertical Forces

- Vertical Forces:

$$\textit{gravitational} = m_E m_{SE} g / (r_E + h)^2$$

$$\textit{centrifugal} = m_{SE} (2\pi/t)^2 (r_E + h)$$

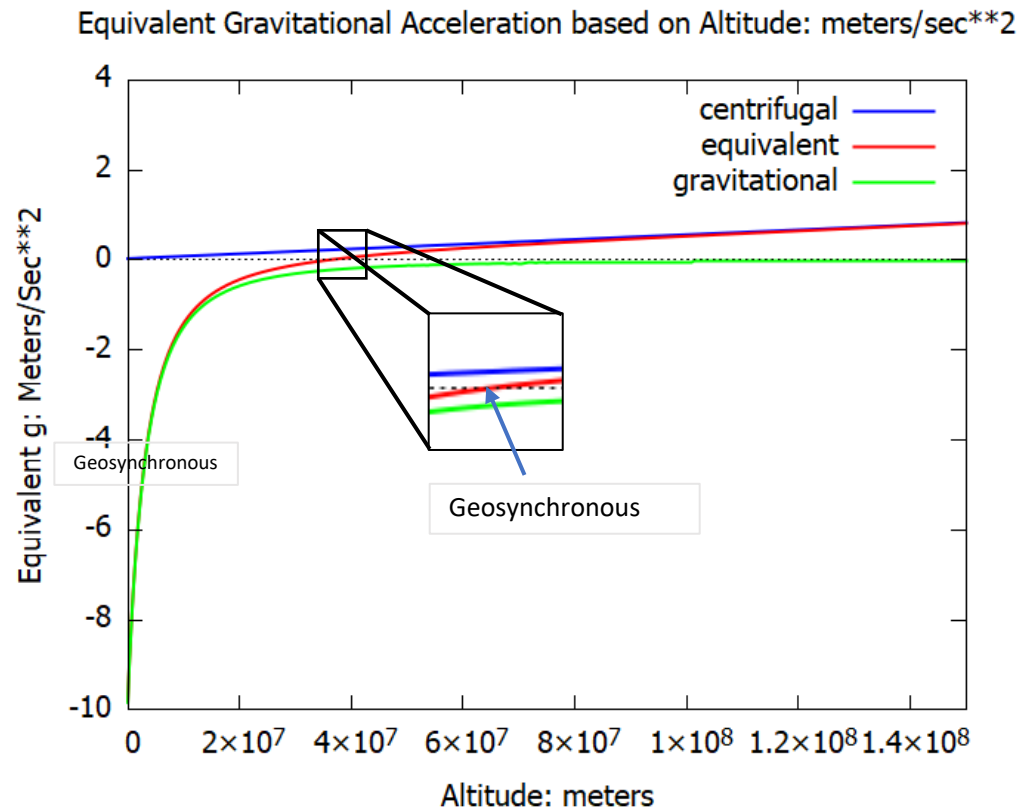
$$\textit{Force Equivalent} = m_{SE} [(2\pi/t_{\textit{revolution}})^2 (r_E + h) - m_E g / (r_E + h)^2]$$

$$\textit{Gravitational Equivalent Acceleration} = [(2\pi/t)^2 (r_E + h) - m_E g / (r_E + h)^2]$$

***Forces are Linear with Mass of Space Elevator***

# Vertical Forces Acting on Space Elevators

- Equivalent Gravitational Acceleration (EGA)



# Energy and Power Require for Vertical Ascent

- Energy & Power Equations:

$$\text{Energy} = (m_{SE}[(2\pi/t_{\text{revolution}})^2(r_E + h) - m_E g/(r_E + h)^2]) dh$$

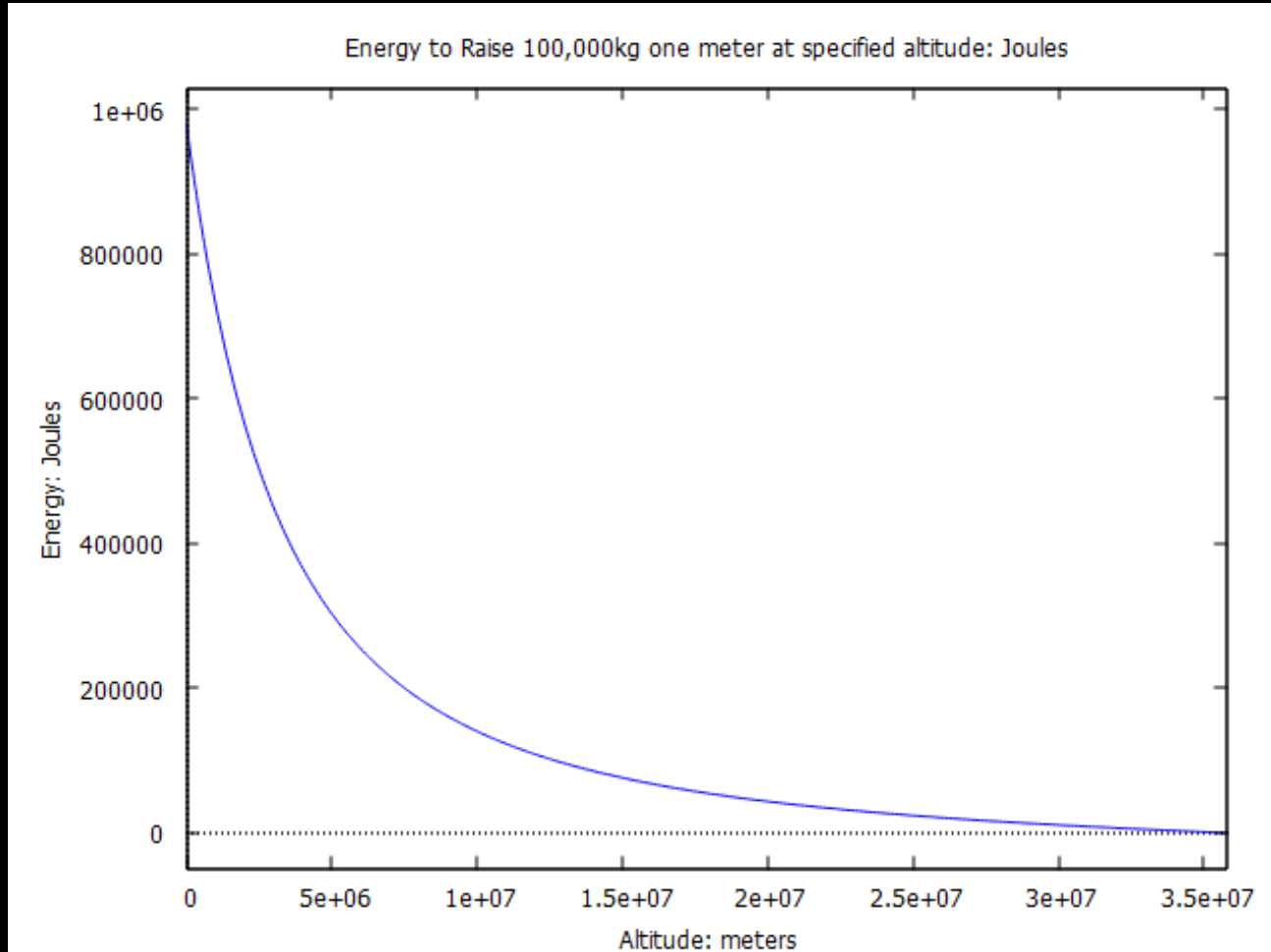
$$\text{Power} = (m_{SE}[(2\pi/t_{\text{revolution}})^2(r_E + h) - m_E g/(r_E + h)^2]) dh/dt$$

$$\text{Accumulated Energy} = \int_{h=0}^{h=\text{max}} (m_{SE}[(2\pi/t_{\text{revolution}})^2(r_E + h) - m_E g/(r_E + h)^2]) dh$$

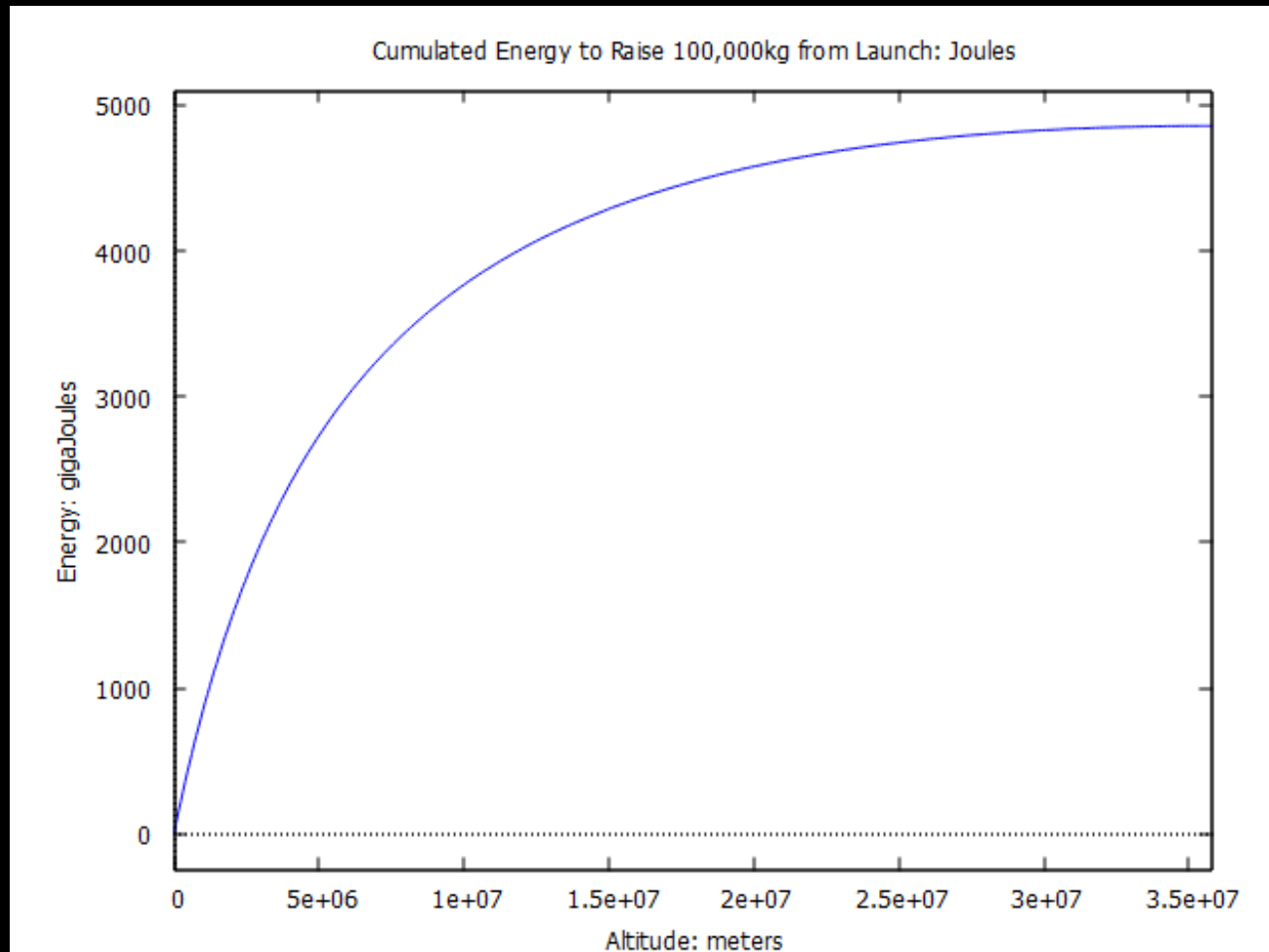
$$\text{Accumulated Power} = \int_{h=0}^{h=\text{max}} (m_{SE}[(2\pi/t_{\text{revolution}})^2(r_E + h) - m_E g/(r_E + h)^2]) dh/dt$$

**Energy & Power are Linear with Mass of Space Elevator**

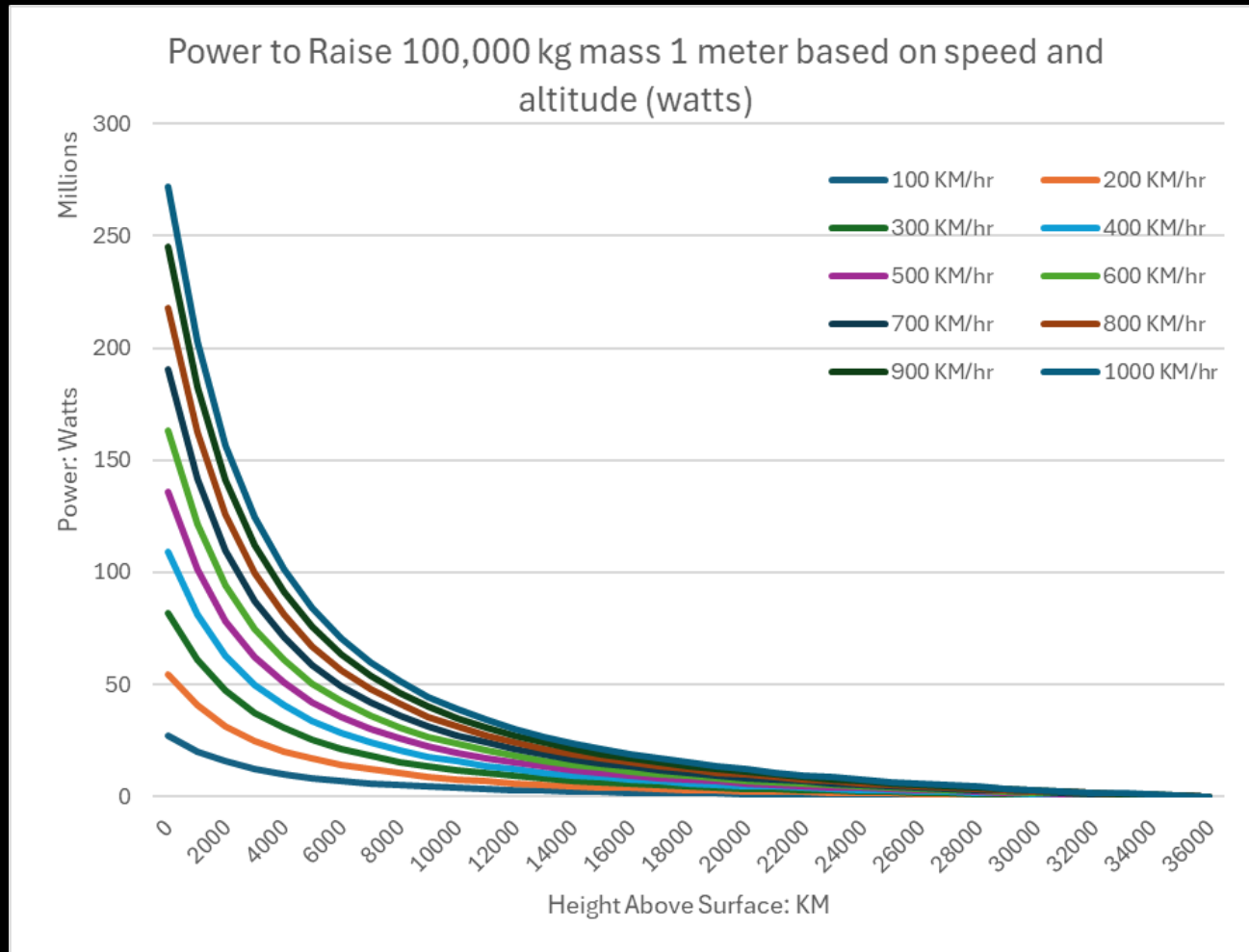
# Energy Required to Raise Mass is Altitude Dependent



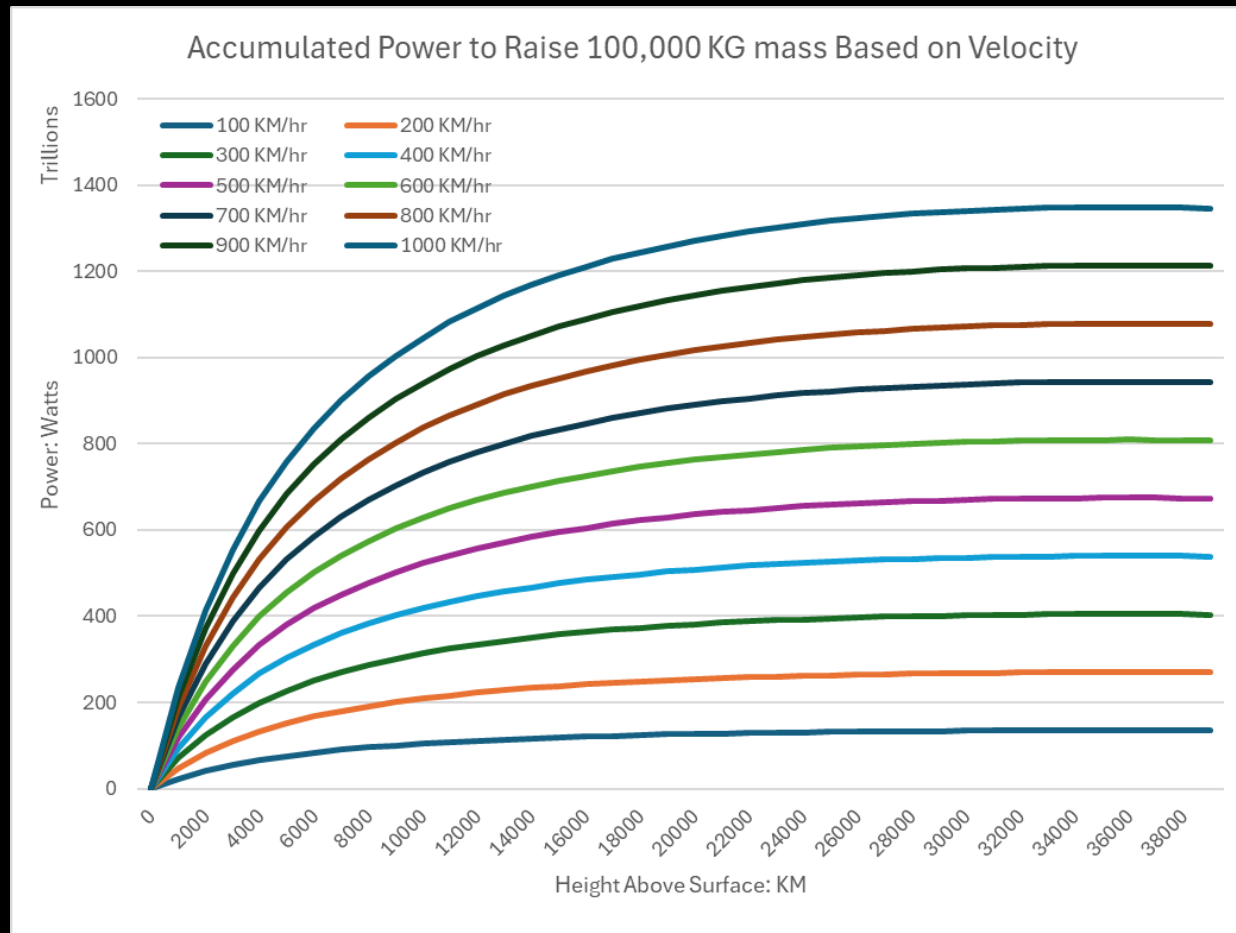
# Accumulated Energy to Raise Mass Drastically Reduces as Its Approaches GEO



# Power To Raise Mass is a Dependent on Altitude and Velocity



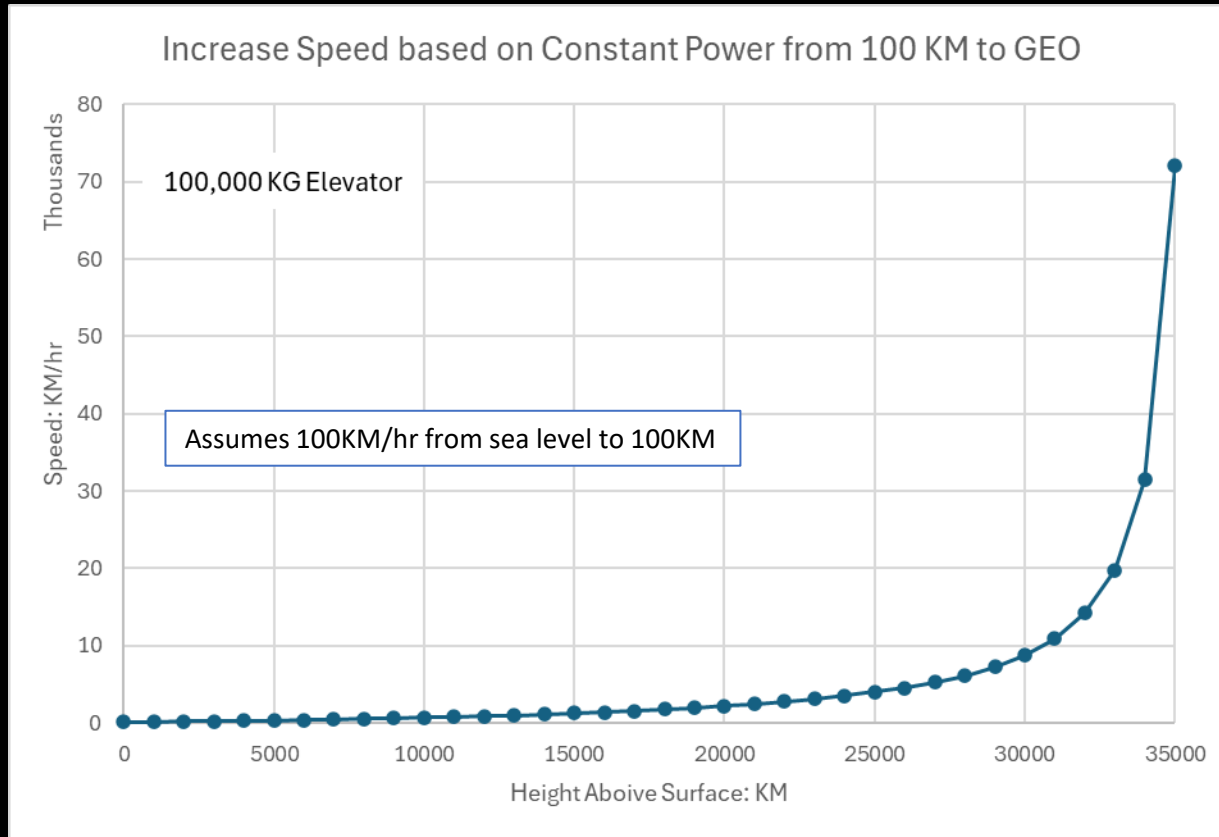
# Accumulated Power to Raise Mass Drastically Reduces as Its Approaches GEO



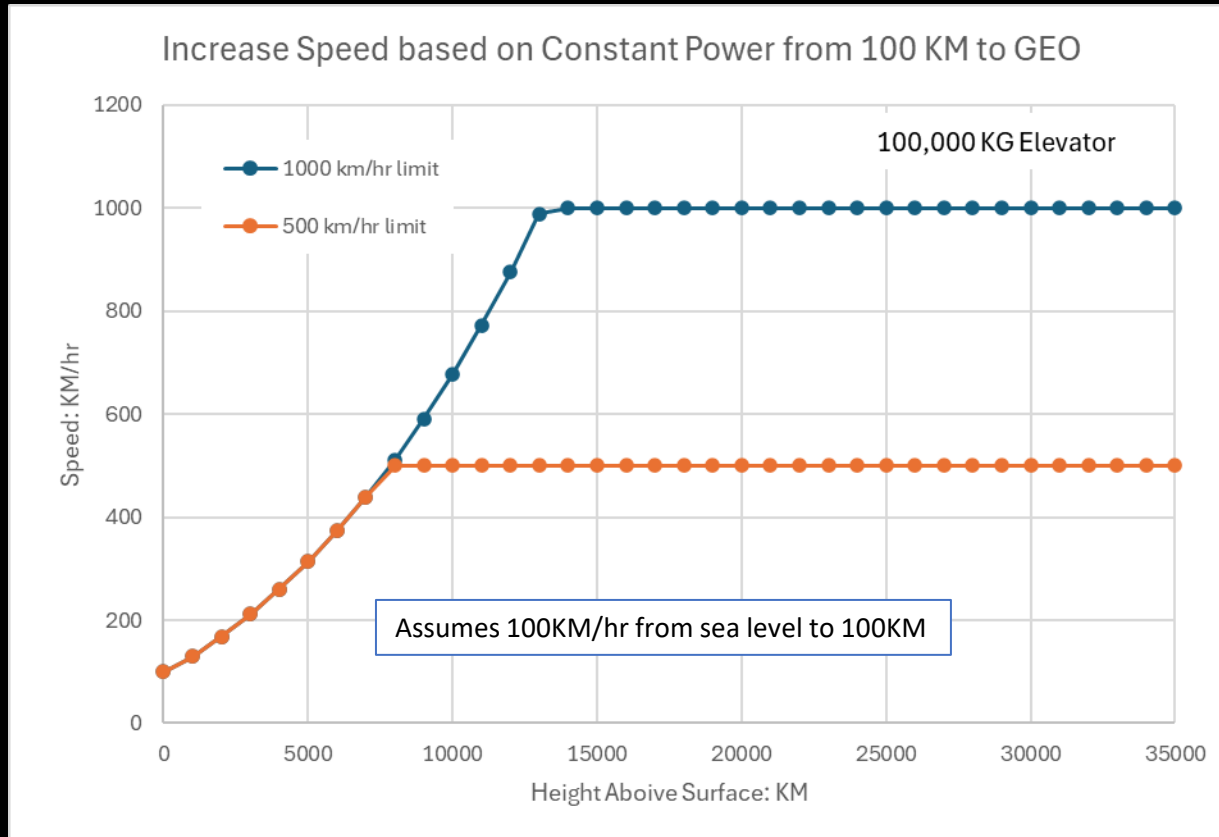
# Mission Considerations

- Time to Reach GEO
- Cargo/Personnel mass required to ascend
- Energy and power for various total vehicle masses and maximum velocities
- Technical and Economic considerations for a single ascent/descent
- Technical and Economic considerations for a Mission's ascent/descent
  - Moon settlement
  - Mars settlement
  - Space solar power
  - Asteroid mining supplies and returns precious metals to Earth
  - Total Space Economy support in transportation needs
  - Cargo & living creatures transportation differences

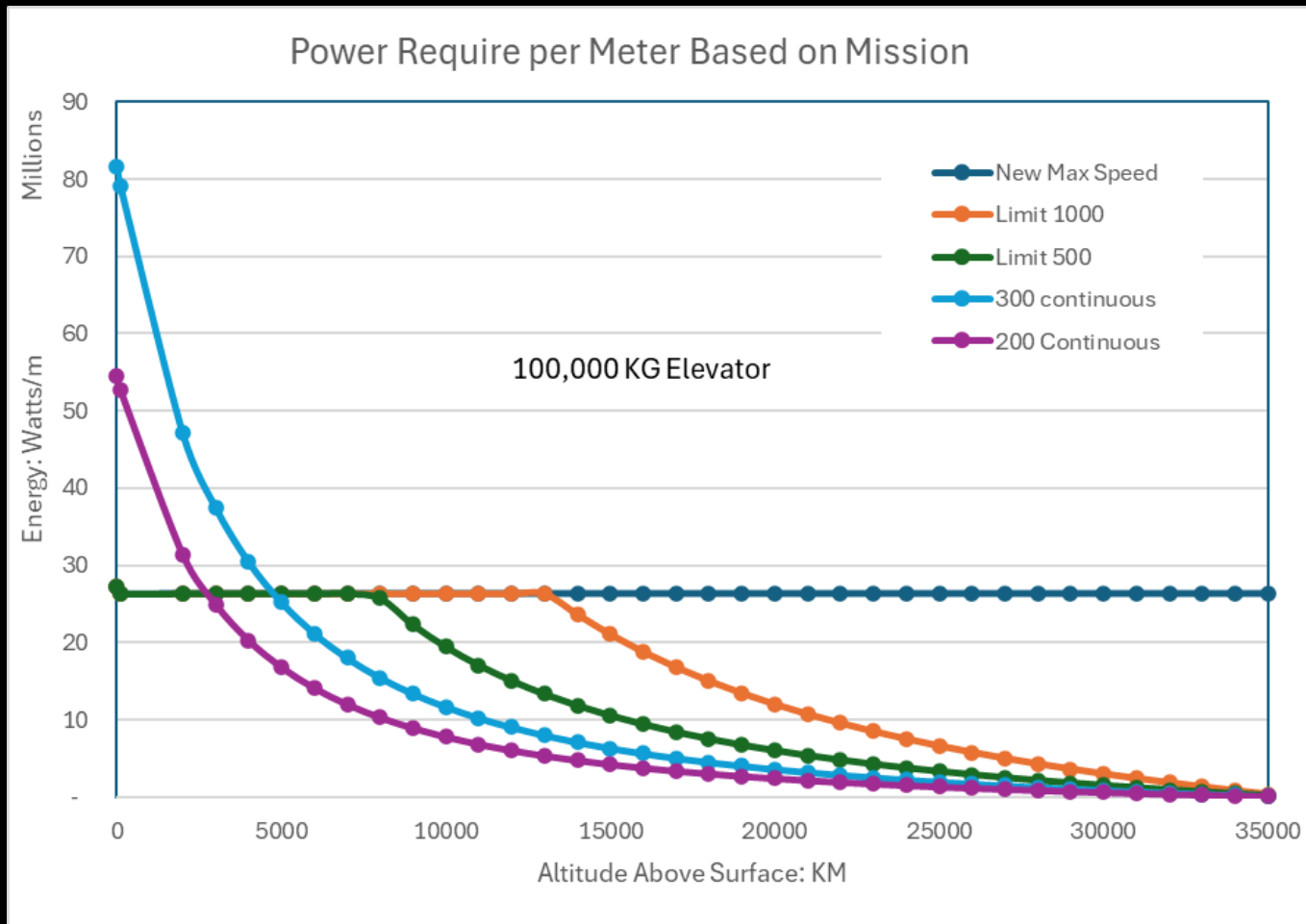
# Constant Power Consumption Allows for Increased Speed



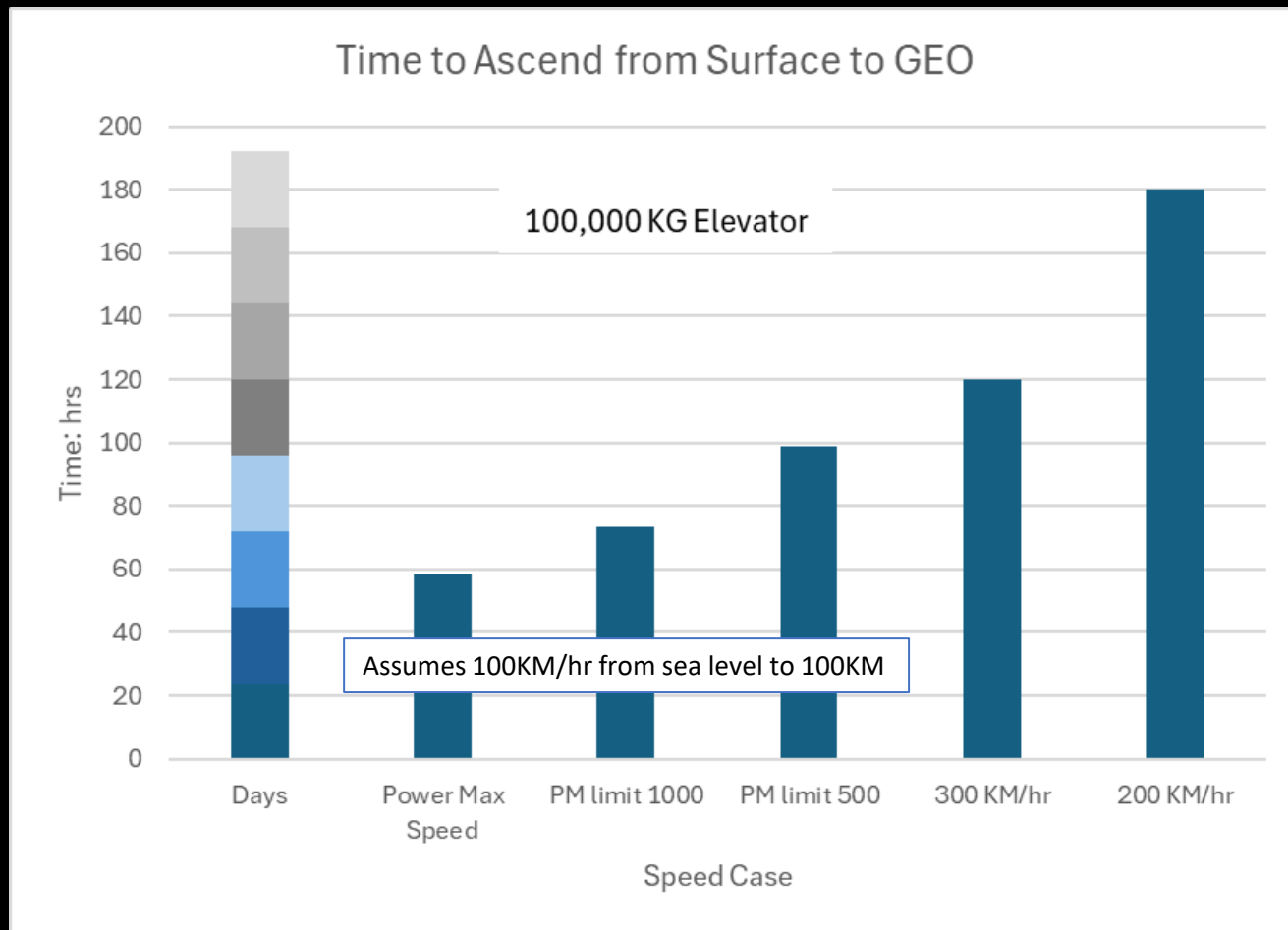
# Constant Power / Maximum Velocity Limitation Profile



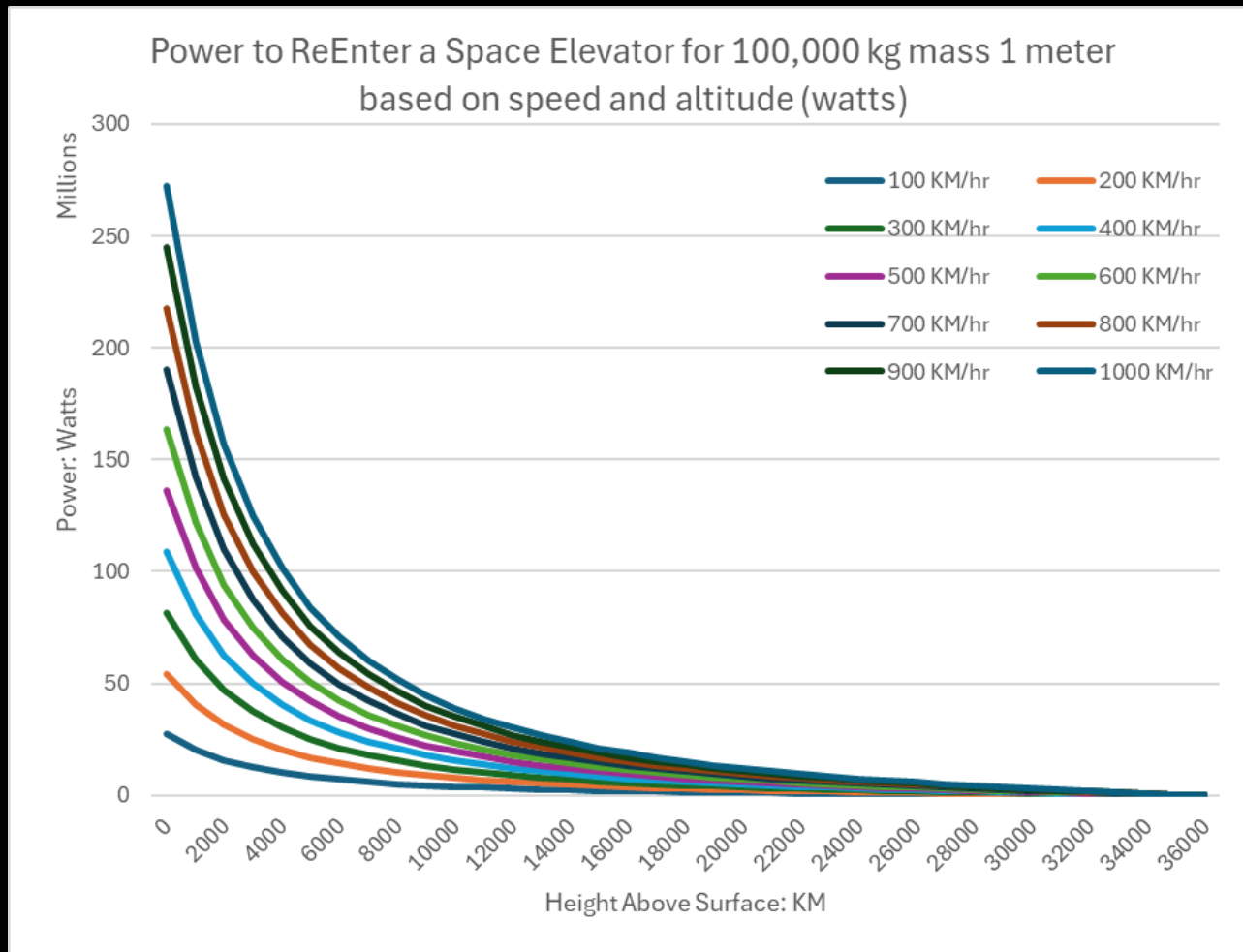
# Power Required for Various Mission Profiles



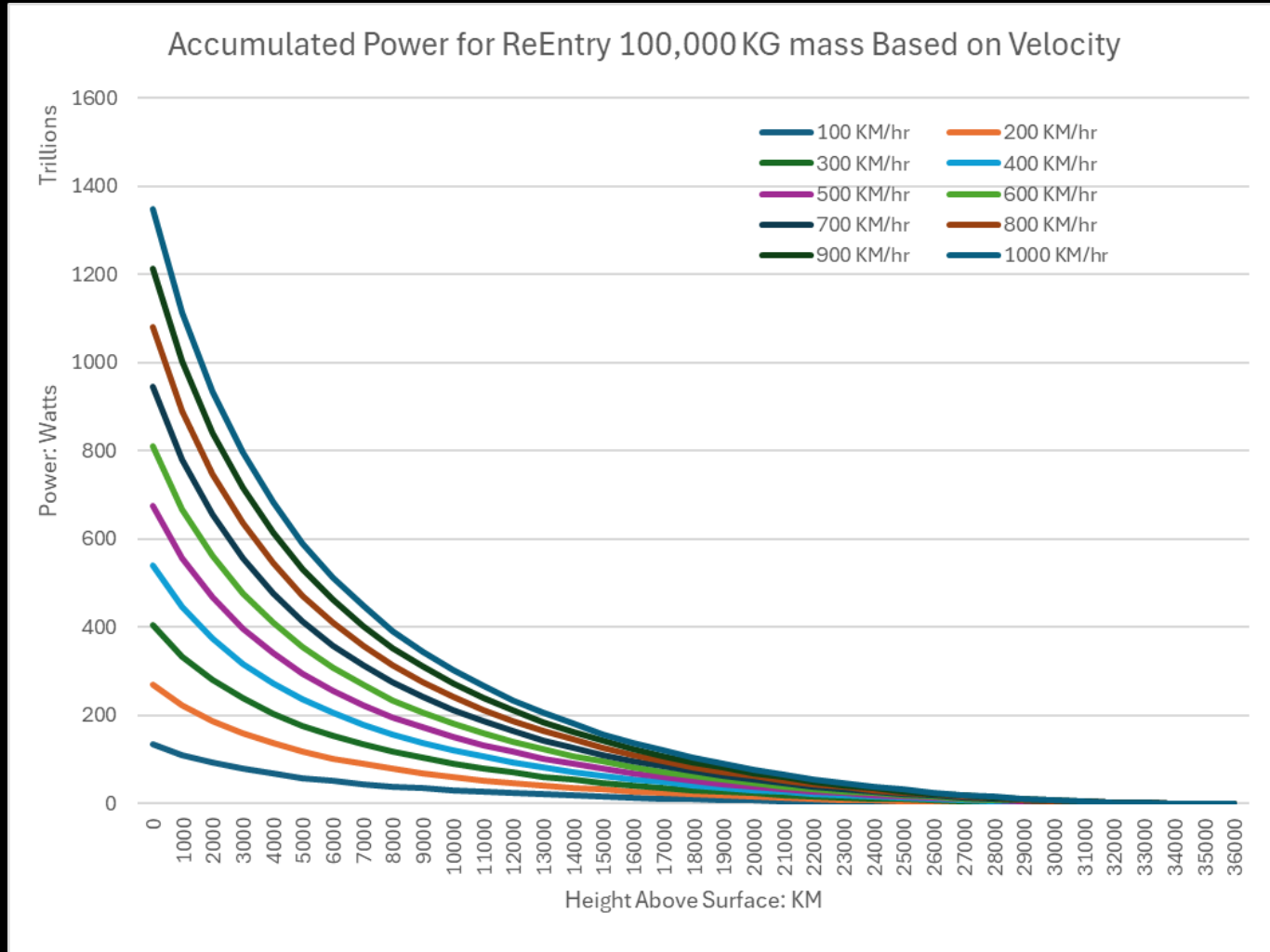
# Maximum Velocity Limitation Profile Ascent Timelines



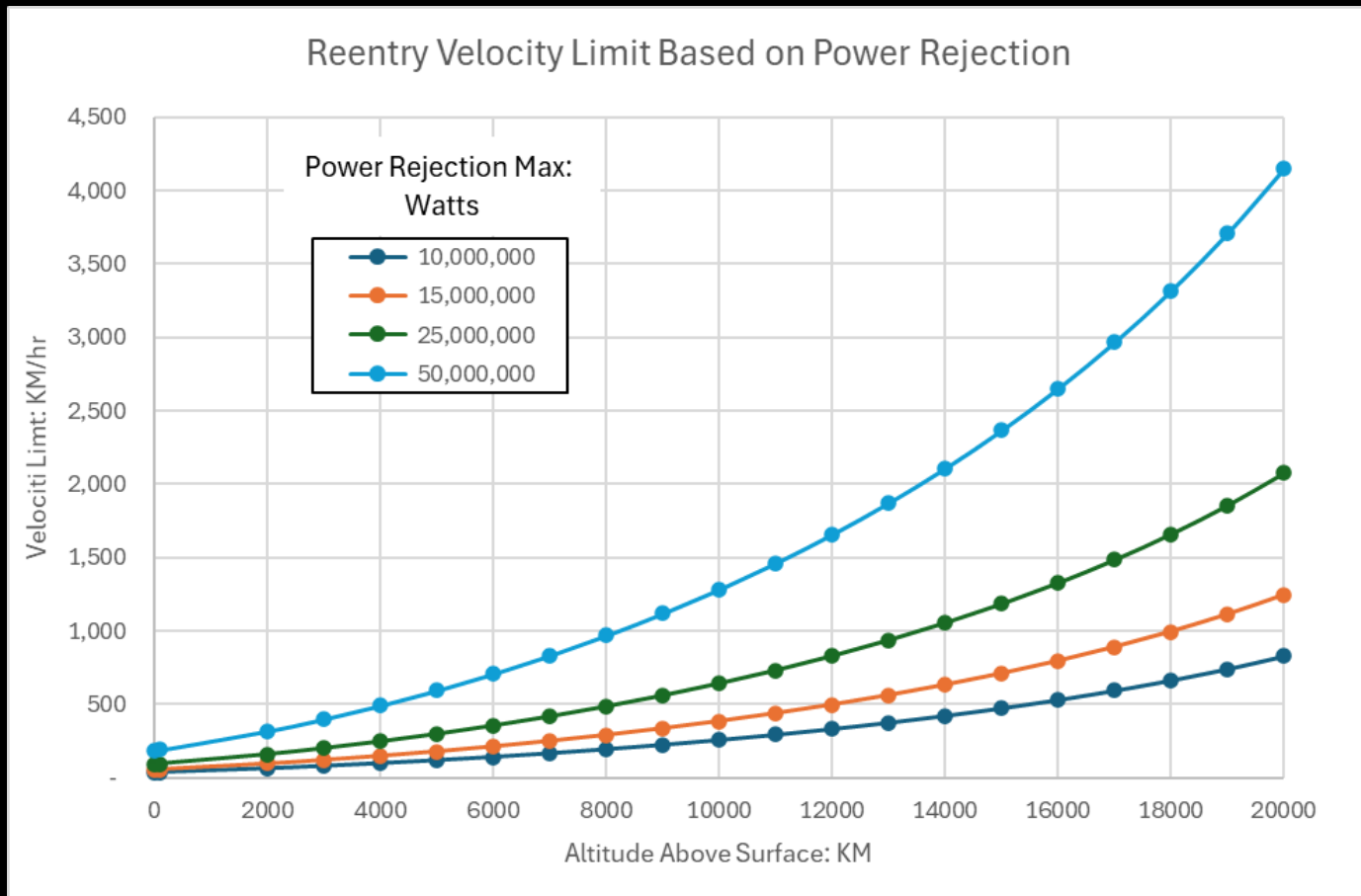
# Power To Absorb Upon Reentry of a Space Elevator is Dependent on Altitude and Velocity



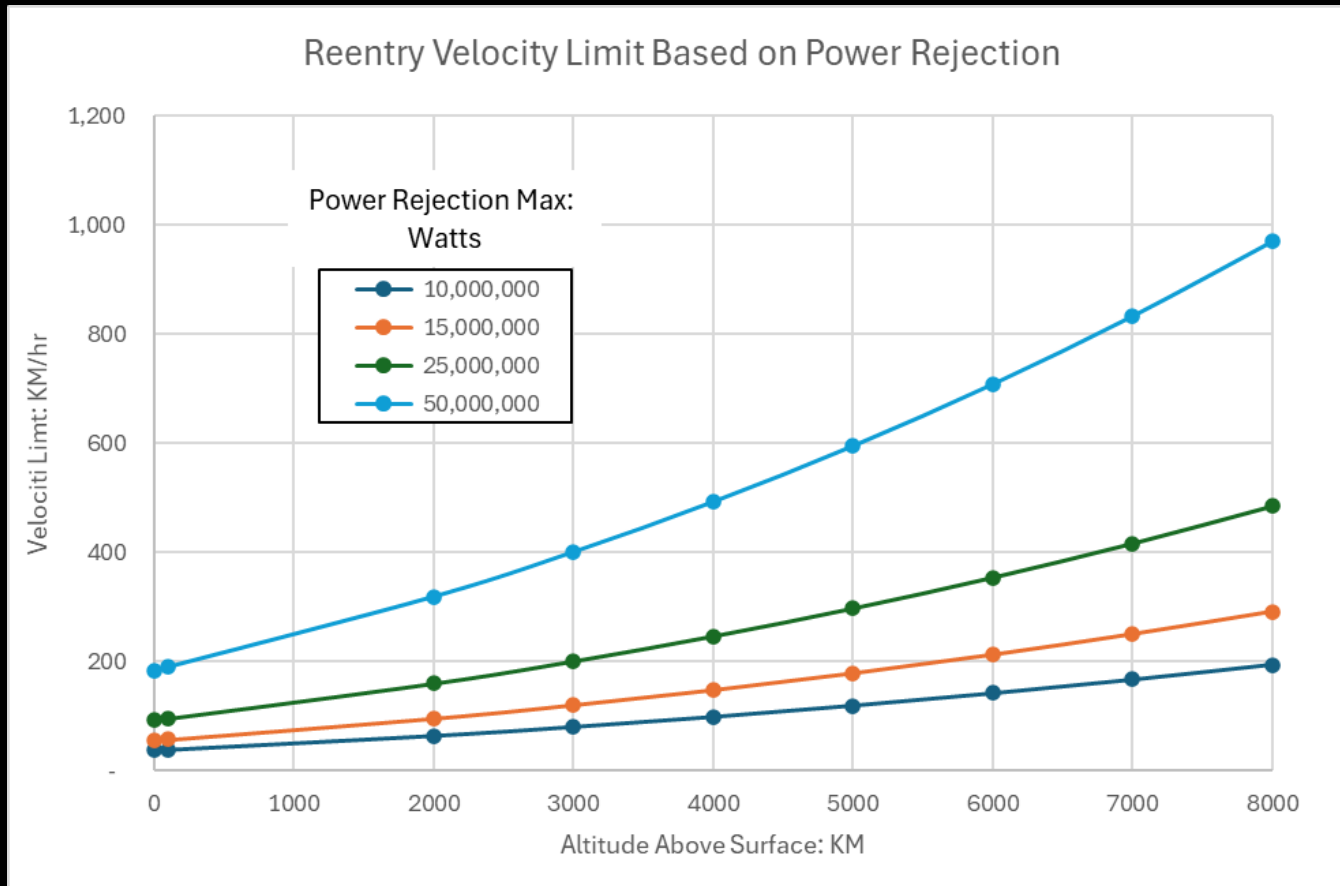
# Reentry Mission Reentry Power Absorption / Transmission Concern



# Reentry Velocity Limits Based on Max Power Rejection



# Reentry Velocity Limits Based on Max Power Rejection



# Initial Assessment of Energy & Power

- Energy & Power per Space Elevator Dependent on:
  - Mass of cargo/personnel
  - Empty mass of space elevator
  - Velocity/velocity vs altitude profile
  - Time limit to GEO, or GEO to surface
- Architecture definition will be based on:
  - Current state of the art for energy & power
  - Expectations of future timeline energy & power capabilities
- Science and Technology development needs
  - Mission needs will drive overall space elevator capability, which will drive overall architecture
  - Overall system architecture will drive science & technology development

# Technical Challenge to Space Elevator Community

- Down-Earth gravitational forces generate considerable regenerative electric power
  - How to manage the electric power generated (use, store, or transmit)
- Inefficiencies in electric power system can create considerable waste heat that must be managed
- Safety issues with down-Earth failure modes
- What is the best vehicle design for transporting tons of precious metals
- What new technologies are needed

# Contact Info

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